

AD-A114 298

AKRON UNIV OH INST OF POLYMER SCIENCE  
CRACK VELOCITIES IN NATURAL RUBBER.(U)  
MAY 82 A N GENT. P MARTENY

F/G 11/10

UNCLASSIFIED

TR-17

N00014-76-C-0408

NL

1 OF 1  
OF 2142 08

END  
DATE  
FILMED  
6 82  
DTIC

12

OFFICE OF NAVAL RESEARCH

Contract N00014-76-C-0408

Project NR 092-555

Technical Report No. 17

CRACK VELOCITIES IN NATURAL RUBBER

by

A. N. Gent and P. Marteny

Institute of Polymer Science  
The University of Akron  
Akron, Ohio 44325

May, 1982

Reproduction in whole or in part is permitted  
for any purpose of the United States Government

Approved for Public Release; Distribution Unrestricted

DTIC FILE COPY

DTIC  
ELECTE  
MAY 10 1982  
S D  
E

82 05 10 080

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER Technical Report No. 17	2. GOVT ACCESSION NO. AD-A144 198	3. RECIPIENT'S CATALOG NUMBER
4. TITLE (and Subtitle) Crack Velocities in Natural Rubber	5. TYPE OF REPORT & PERIOD COVERED Technical Report	
	6. PERFORMING ORG. REPORT NUMBER	
7. AUTHOR(s) A. N. Gent and P. Marteny	8. CONTRACT OR GRANT NUMBER(s) N00014-76-C-0408	
9. PERFORMING ORGANIZATION NAME AND ADDRESS Institute of Polymer Science The University of Akron Akron, Ohio 44325	10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS NR 092-555	
11. CONTROLLING OFFICE NAME AND ADDRESS Office of Naval Research Power Program Arlington, VA 22217	12. REPORT DATE May, 1982	
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office)	13. NUMBER OF PAGES 22	
	15. SECURITY CLASS. (of this report) Unclassified	
15a. DECLASSIFICATION/DOWNGRADING SCHEDULE		
16. DISTRIBUTION STATEMENT (of this Report) According to attached distribution list Approved for public release; distribution unrestricted.		
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report)		
18. SUPPLEMENTARY NOTES Submitted for publication in: Journal of Materials Science		
19. KEY WORDS (Continue on reverse side if necessary and identify by block number) Biaxial strains, Rubber, Crack velocity, Fracture, Strain, Velocity of sound.		
20. ABSTRACT (Continue on reverse side if necessary and identify by block number) Measurements are reported of the limiting velocity of a running crack in biaxially stretched sheets of unfilled and carbon-black-filled natural rubber. The crack velocity was found to increase with the cleavage strain $\epsilon_y$ and also with the strain $\epsilon_x$ parallel to the direction of tearing, reaching values of over 100 m/s at the highest strain levels employed. These crack velocities are shown to be close to those predicted by Mott's theory, i.e., about one-third of the velocity of		

DD FORM 1473  
1 JAN 73

EDITION OF 1 NOV 65 IS OBSOLETE  
S/N 0102-LF-014-6601

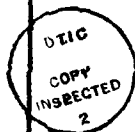
SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

20. (continued)

sound, when a strong strain-dependence is recognized for the velocity of sound in rubber.

Accession For	
NTIS GRA&I	<input checked="" type="checkbox"/>
DTIC TAB	<input type="checkbox"/>
Unannounced	<input type="checkbox"/>
Justification	
By	
Distribution/	
Availability Codes	
Dist	Avail and/or Special
A	



S/N 0102- LF- 014- 6601

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

## I. Introduction

The maximum velocity  $\underline{v_c}$  of a running crack has been shown by Mott to be related to the velocity  $\underline{v_s}$  of sound (1). Roberts and Wells computed the ratio  $\underline{v_c}/\underline{v_s}$  to be 0.38 for a material having a value of Poisson's ratio of 0.25(2). Reported values of  $\underline{v_c}$  and  $\underline{v_s}$ , although relatively few in number, are in reasonably good agreement with this prediction.

For highly-elastic materials, however, the agreement is less satisfactory. Mason found  $\underline{v_c}/\underline{v_s}$  to be about 0.3 for a vulcanized SBR elastomer and about 0.03 for a vulcanized natural rubber (3). The surprisingly low value for natural rubber was attributed to highly anisotropic elastic behavior at high strains, rendering invalid the theoretical treatments of Mott, and Roberts and Wells, which assumed small-strain isotropic behavior. Recent studies by Stevenson and Thomas of the velocities of crack propagation in bursting rubber balloons led to estimates of the ratio  $\underline{v_c}/\underline{v_s}$  of about 3.2 for natural rubber, about two orders of magnitude greater than Mason's result (4).

No other studies of high-speed crack velocities in rubber are known to the present authors. Measurements have therefore been carried out for carbon-black-filled and unfilled natural rubber sheets, held under various

states of strain. They are reported here and compared with recent measurements of the velocity  $v_s$  of sound in stretched rubber (5).

## 2. Experimental details

### (i) Materials used

Rubber sheets were prepared using the mix formulations and vulcanization conditions given in the Appendix. Test strips were cut from the sheets about 250 mm long, 20 mm wide and 0.5 mm thick. Some samples of similar dimensions were prepared in a special mold, to give thickened edges along the 250 mm sides for ease of clamping. A grid was painted on one surface of the test strips to allow strains to be measured.

### (ii) Biaxial straining

The test strip was first stretched along its length, denoted the  $x$  direction, to the required strain  $e_x$ . Two parallel rigid clamps, 150 mm long, were then applied to the strip edges so that a section of the strip, about 150 mm long and 10 mm wide, was secured between them in the stretched state. When these clamps were fastened, the original stretching force was released. The specimen then remained stretched in its length direction (except for small regions at the ends), because the clamps along each

side prevented it from returning to the unstrained state.

The clamps were then attached to a loading device, arranged to pull them apart in the  $y$  direction and thus impose on the rubber strip an additional strain  $e_y$  in a direction perpendicular to the first. The strain  $e_y$  was always made greater than the strain  $e_x$  so that a crack initiated in the center of the strip would run in the long  $x$ -direction rather than in the short  $y$ -direction and would therefore be more easily studied.

Measurements were made of the amount of work required to stretch the rubber sheet in the  $y$ -direction to the strain  $e_y$ . On dividing by the volume of rubber, this gave the strain energy density  $W$  for a particular combination of strains  $e_x$ ,  $e_y$  that would be released by a crack running in the  $x$ -direction. (It is assumed that the original work of imposing the strain  $e_x$  is not released by such a crack). The amount of energy released by such a crack, per unit area of material torn through, is then given by

$$T = Wh_o \quad (1)$$

where  $h_o$  denotes the strip width in the  $y$  direction between the clamps, when the force in this direction is

zero (6). The tear energy  $\underline{T}$  has been shown to be the parameter governing slow crack growth in elastomeric materials under varied loading conditions (7-10). Cracks do not propagate unless  $\underline{T}$  exceeds a well-defined critical value. Above this value, the rate of propagation appears to depend solely upon the magnitude of  $\underline{T}$ . The same concept has been applied to other materials.  $\underline{T}$  is termed the strain energy release rate and its critical value, sometimes denoted  $\underline{G}_c$ , is termed the fracture energy of the material. We examine below whether the magnitude of  $\underline{T}$  governs the velocity of high-speed cracks.

(iii) Measuring the crack velocity

A crack was initiated in the center of a biaxially-stretched strip by piercing it with a needle point that had been filed to resemble a spearhead in order to help guide the crack in the  $\underline{x}$ -direction. As the crack grew it permitted a light beam from a laser source to illuminate a photo-sensitive trigger and thus to set off a series of three flashtubes, Figure 1, giving a triple exposure of the propagating crack on a photographic film. The same arrangement was employed previously for determining velocities of free retraction of stretched rubber strips, and has been described more fully in that connection (5).



A representative photograph is shown in Figure 2. From the distance moved by the crack tip between successive exposures, and the measured time interval between flashes, mean values of the velocity  $\underline{v}_c$  of crack propagation were determined. No indication was found either of acceleration or deceleration of the crack over the distances studied, about 5 cm of growth on each side. This conclusion is supported by the observations of Mason (3) and Stevenson and Thomas (4).

### 3. Results and discussion

- (i) The relation between crack velocity and tear energy

As will be discussed in more detail later, measured crack velocities were found to depend strongly upon the levels of strain  $\underline{e}_x$  and  $\underline{e}_y$ . If  $\underline{e}_y$  was not sufficiently large, then the crack did not propagate at all. Above the critical value of  $\underline{e}_y$ , the crack grew at rates between about 5 and 100 m/s, depending upon the strains imposed. The first question to consider, then, is whether the crack velocity depends solely upon the available energy  $\underline{T}$ , or whether it is a function of the local state of strain only. These parameters can be adjusted separately with the specimens used in the present experiments. By varying the width  $\underline{h}_o$  of the strip between the clamps, the magnitude of  $\underline{T}$

can be changed even when the strain levels and the strain energy density  $\underline{W}$  are held constant, equation 1.

Measurements were made of the crack velocity  $\underline{v}_c$  in strips of unfilled natural rubber of various widths  $\underline{h}_0$  ranging between 2 and 15 mm, when the strips were stretched to  $\underline{e}_x = 1$  and  $\underline{e}_y = 3$ . At these strain levels,  $\underline{W} = 3 \text{ MJ/m}^3$ . The results are shown in Figure 3. Until  $\underline{h}_0 = 3 \text{ mm}$ , the crack velocity was effectively zero. From equation 1, the critical tear energy  $\underline{T}$  for a crack to propagate at all is thus about  $9 \text{ kJ/m}^2$ . Measurements at other strain levels confirmed the validity of this tear energy criterion for any crack growth to occur. Above this energy level, as shown in Figure 3, the crack velocity increased rapidly with increasing width  $\underline{h}_0$  of the strip, to reach an upper value of about 53 m/s. It then became quite independent of the strip width, i.e., of the tear energy  $\underline{T}$ . Thus, it may be concluded that whereas a tear energy criterion governs the onset of fracture and probably governs the rate of slow-speed cracks, the maximum crack velocity is independent of the available energy for tearing. It depends strongly upon the imposed strains, however, as discussed below.

(ii) Effect of imposed strains upon the maximum crack velocity

Measured crack velocities  $\underline{v}_c$  are plotted as a function of the imposed strains in Figures 4 and 5. For the unfilled material A,  $\underline{v}_c$  increases with increasing strain  $\underline{e}_y$ . Moreover, at any value of  $\underline{e}_y$ ,  $\underline{v}_c$  increases markedly with  $\underline{e}_x$ . Apparently the material tears much more rapidly when it is held somewhat stretched in the direction of the running crack. A pronounced reduction in tear strength for a sample held stretched in the tear direction has been noted previously (11).

Similar behavior was shown by the carbon-black-filled material B, Figure 5. However, relatively slow crack growth was observed for this material over a range of strains  $\underline{e}_y$  from about 1.2 to 2.0, when the strain  $\underline{e}_x$  was small or zero. Over this range of strains the crack velocity was anomalously low, only about 3 m/s. On examining the torn surfaces they were relatively smooth and structureless at a magnification of 300x, for all strain conditions, whether the crack had grown at about 3 m/s or at a much higher velocity. The anomalously-low rates of tearing at low strains cannot therefore be ascribed to a change in the geometry of tearing and must reflect an intrinsic resistance to fracture at high speeds, presumably from an energy-dissipation mechanism in material B that is not present

in material A. One such process would be the detachment of rubber from filler particles under the high stresses set up at the crack tip.

(iii) Comparison with Mott's theory

As shown in Figures 4 and 5 the crack velocity increased rapidly with increasing strain for both the unfilled and carbon-black-filled materials, reaching values of over 100 m/s at the highest strain levels employed. In experiments with a bursting balloon, where the strain  $e_y (=e_x)$  ranged from 1.3 to 4.0 strain units, Stevenson and Thomas reported even higher velocities  $v_c$  of 130 to 285 m/s (4). These values are now compared with the predictions of Mott's theory.

By including kinetic energy in the relation for energy changes during propagation of a crack, Mott showed that the maximum crack velocity should be proportional to the velocity  $v_s$  of a stress pulse in the material (1). Roberts and Wells determined the constant of proportionality to be a rather complicated function of Poisson's ratio, taking the approximate value of 0.38 when Poisson's ratio is 0.25 (2). As described elsewhere, this constant becomes 0.3 when Poisson's ratio is taken as 0.5, the appropriate value for virtually incompressible materials like rubber (6).

Now the velocity  $\underline{v_s}$  of a stress pulse in rubber is strongly dependent upon the state of strain (5). Some representative values are given in Table 1. Values of the maximum crack velocity  $\underline{v_c}$ , calculated from measured values of  $\underline{v_s}$  on the assumption that Mott's relation holds for stretched rubber, are given in Table 1. They are also shown in Figures 4 and 5 as full curves, for comparison with the experimentally-determined values of  $\underline{v_c}$ . They are seen to be in reasonably good agreement with the highest crack velocities measured, i.e., those observed when the strains  $\underline{e_x}$  and  $\underline{e_y}$  were both relatively large. Moreover, the measured values increase rapidly with the imposed strain  $\underline{e_y}$ , in good agreement with the predicted relations.

Thus, for both unfilled and carbon-black-filled samples of natural rubber, the maximum observed velocities of crack propagation are in good agreement with the relation

$$\underline{v_c} = 0.3 \underline{v_s} \quad (2)$$

when the velocity  $\underline{v_s}$  is determined at an appropriate state of strain. Because  $\underline{v_s}$  is strongly dependent upon the state of strain, the maximum crack velocity is predicted to increase from about 20 m/s to about 200 m/s for the unfilled material A over the range of

strains  $\underline{e}_y = 1$  to  $\underline{e}_y = 4.5$ , and from about 90 m/s to about 220 m/s for the carbon-black-filled material B over the range of strains  $\underline{e}_y = 1$  to  $\underline{e}_y = 2$ . The measured values are in reasonably good agreement with these predictions.

It should be noted that no allowance has been made for possible changes in the velocity  $\underline{v}_s$  when a strain  $\underline{e}_x$  is imposed perpendicular to the direction of travel of the stress pulse. The predicted relations are based solely on values of  $\underline{v}_s$  determined for strips in a state of simple extension, i.e., with  $\underline{e}_x = \underline{e}_y^{-\frac{1}{2}}$ . A large effect of strains  $\underline{e}_x$  imposed in a perpendicular direction is not expected for the velocities  $\underline{v}_s$ , however.

(iv) Comparison with other work

Crack tip velocities reported by Stevenson and Thomas (4) are given in Table 1. They are seen to be in reasonable agreement with values calculated for a carbon-black-filled natural rubber material B, using the measured velocity of sound at the particular imposed strain. As Stevenson and Thomas pointed out, the velocity of sound at small strains is clearly far too low to account for their observed crack velocities and those reported here. They were not aware that much higher sound velocities are encountered at moderately high strains, and attributed the high crack velocities

to special conditions set up at the crack tip itself. This assumption does not now seem to be necessary because the observed crack velocities can be accounted for solely in terms of the measured strain dependence of the velocity of a stress pulse in rubber.

4. Conclusions

The velocity of crack propagation in biaxially-strained sheets of natural rubber is an increasing function of the biaxial strains  $\underline{e}_x$  and  $\underline{e}_y$ . Below a critical value of the cleavage strain  $\underline{e}_y$ , which depends on the strip width, no crack growth occurred. These critical conditions correspond to an energy requirement for fracture of about  $5-10 \text{ kJ/m}^2$  for an unfilled natural rubber vulcanizate A and about  $15-20 \text{ J/m}^2$  for a carbon-black-filled natural rubber vulcanizate B.

For the filled material, cracks grew at relatively low velocities, about 3 m/s for a range of imposed strains above the critical level and then the velocity increased rapidly at higher strains. In contrast, cracks grew rapidly in the unfilled material once the cleavage strain was high enough to permit tearing to occur at all. The velocities of high-speed cracks were found to be independent of the available energy for fracture and depended solely upon the state of strain.

Values of maximum velocity  $v_c$  were calculated from measured values of the velocity  $v_s$  of sound at various imposed strains. Agreement with the observed maximum crack velocities was surprisingly good, both in general magnitude and in the strong dependence upon the imposed strain, for both materials examined.

Acknowledgements

This work was supported in part by a research grant from the Office of Naval Research (Contract ONR N00014-76-C-0408) and in part by a research grant from Lord Kinematics Division of Lord Corporation.



References

1. N. F. Mott, Engineering, 165, 16 (1948).
2. D. K. Roberts and A. A. Wells, Engineering, 178, 820 (1954).
3. P. Mason, J. Appl. Phys., 29, 1145 (1958).
4. A. Stevenson and A. G. Thomas, J. Phys. D: Appl. Phys, 12, 2101 (1979).
5. A. N. Gent and P. Marteny, submitted to J. Appl. Phys.
6. P. Marteny, submitted to International J. Fracture.
7. H. W. Greensmith and A. G. Thomas, J. Polymer Sci., 18, 189 (1955).
8. L. Mullins, Trans. Instn Rubber Industry, 35, 213 (1959).
9. A. G. Thomas, J. Appl. Polymer Sci., 3, 168 (1960).
10. A. Kadir and A. G. Thomas, Chap. 5 in "Elastomers: Criteria for Engineering Design", ed. by C. Hepburn and R. J. W. Reynolds, Applied Science Publishers, London, 1979.
11. A. N. Gent and H. J. Kim, Rubber Chem. Technol., 51, 35 (1978).

Appendix

The mix formulations are given below in parts by weight.

Unfilled natural rubber (A):

Natural rubber (SMR-5L), 100; zinc oxide, 5;  
stearic acid, 2; phenyl-2-naphthylamine, 1;  
N-cyclohexyl-2-benzothiazyl sulfenamide, 0.6;  
sulfur, 2.5.

Carbon-black-filled natural rubber (B):

As for A, with the addition of N330 carbon black  
(Vulcan 3, Cabot Corporation), 50.

The compounds were vulcanized in the form of thin sheets by heating them in a press for 24 min at 150°C.

Table 1. Representative values of the velocity  $\underline{v_s}$  of a stress pulse, taken from reference (5), and crack velocities  $\underline{v_c}$ .

$\underline{e_y}$	$\underline{v_s}$ (m/s)	$\underline{v_c}$ (calc. from equation 2) (m/s)	$\underline{v_c}$ from [reference (4)] (m/s)
Unfilled material A			
1.3	75	22.5	133
2.0	90	27	---
2.7	180	57	230
3.0	250	75	---
4.0	400	120	285
Carbon-black-filled material B			
1.3	380	123	133
2.0	720	216	---
2.7	>900	>270	230
3.0	---	---	---
4.0	---	---	285

Figure Captions

- Figure 1. Method of measuring the velocity of a crack in a biaxially-strained rubber strip.
- Figure 2. Triple-exposure photograph of a growing crack in material B.  $e_x = 0.4$ ,  $e_y = 2.3$ .
- Figure 3. Crack velocity  $v_c$  in strips of varying unstrained width  $h_0$ . Material A;  $e_x = 1$ ,  $e_y = 3$ .
- Figure 4. Crack velocity  $v_c$  in biaxially-strained sheets of material A. The vertical dotted line represents the threshold strain below which the crack did not grow at all. The full curve represents the predictions of Mott's theory:  $v_c = 0.3 v_s$ , using values of  $v_s$  appropriate to the strain  $e_y$ .
- Figure 5. Crack velocity  $v_c$  in biaxially-strained sheets of material B. The vertical dotted line represents the threshold strain below which the crack did not grow at all. The full curve represents the predictions of Mott's theory:  $v_c = 0.3 v_s$ , using values of  $v_s$  appropriate to the strain  $e_y$ .

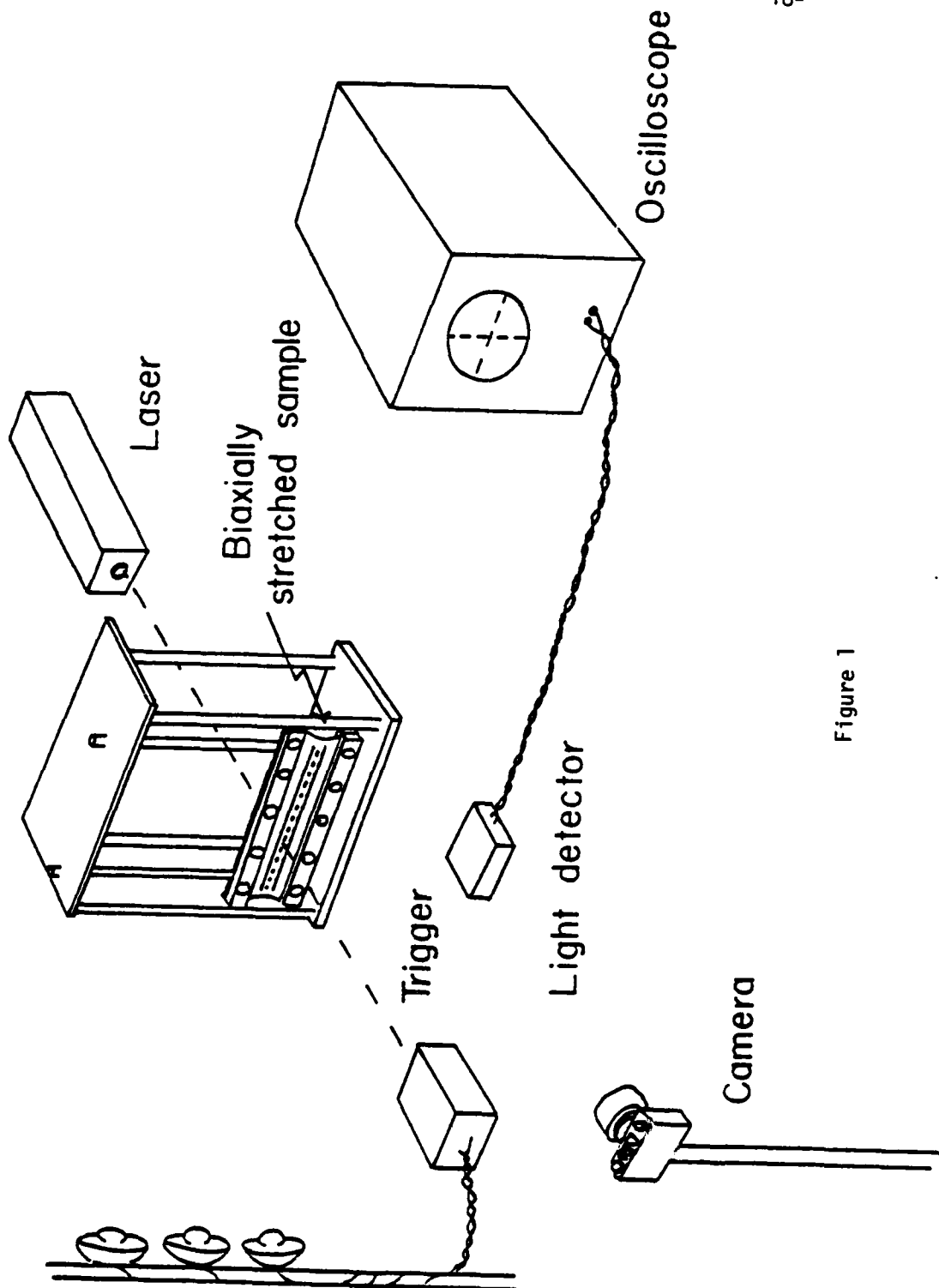


Figure 1

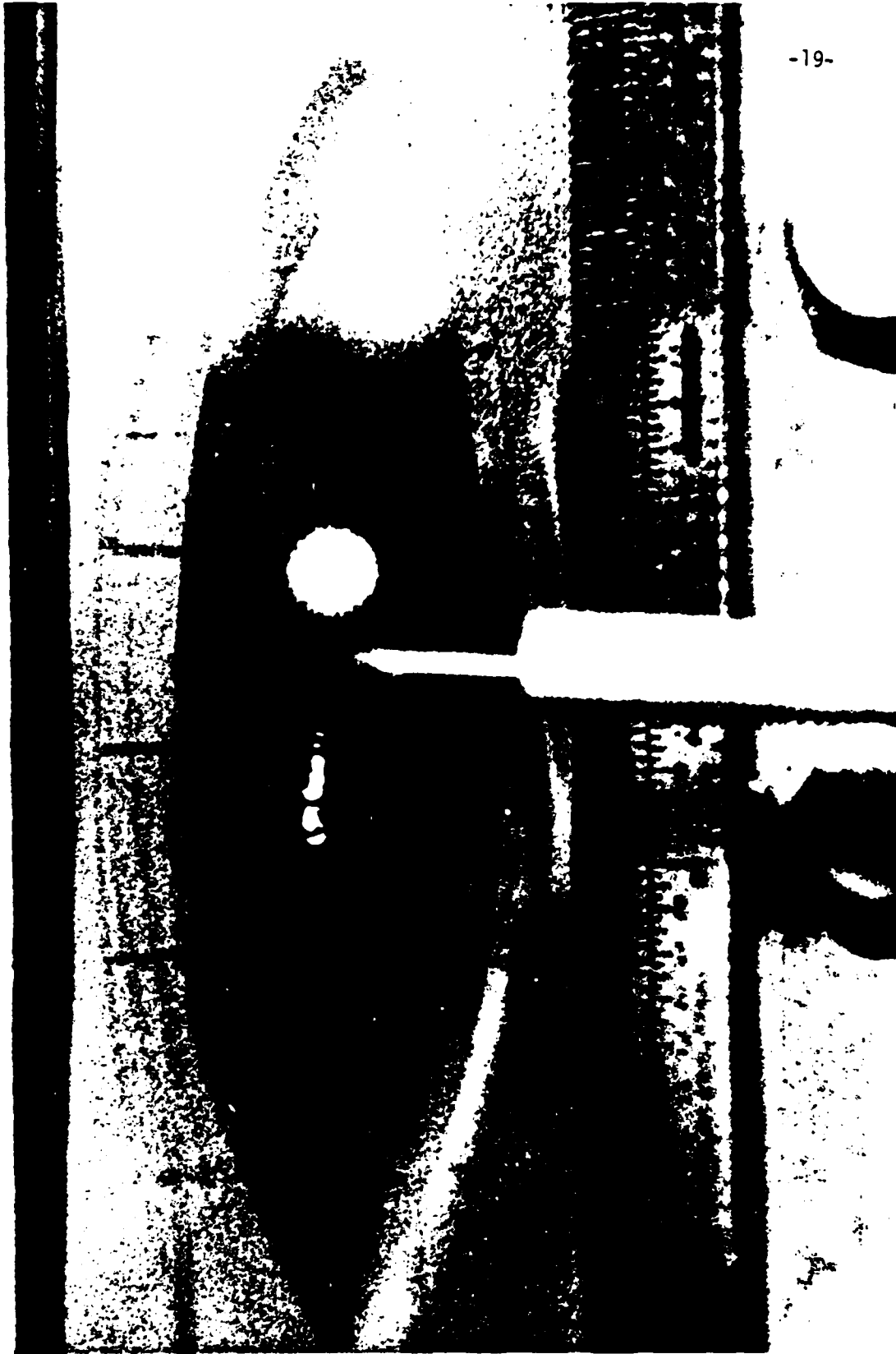


Figure 2

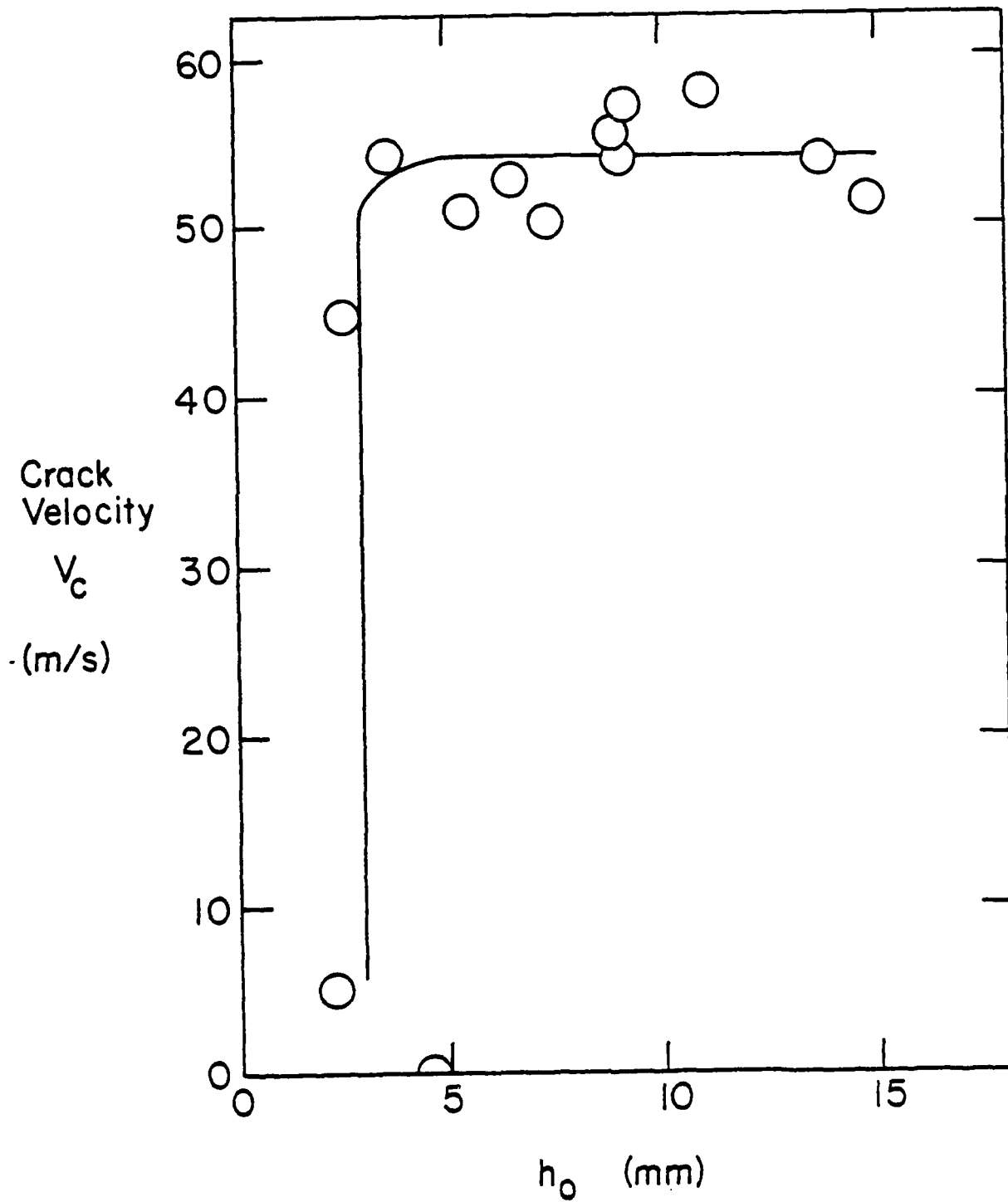


Figure 3

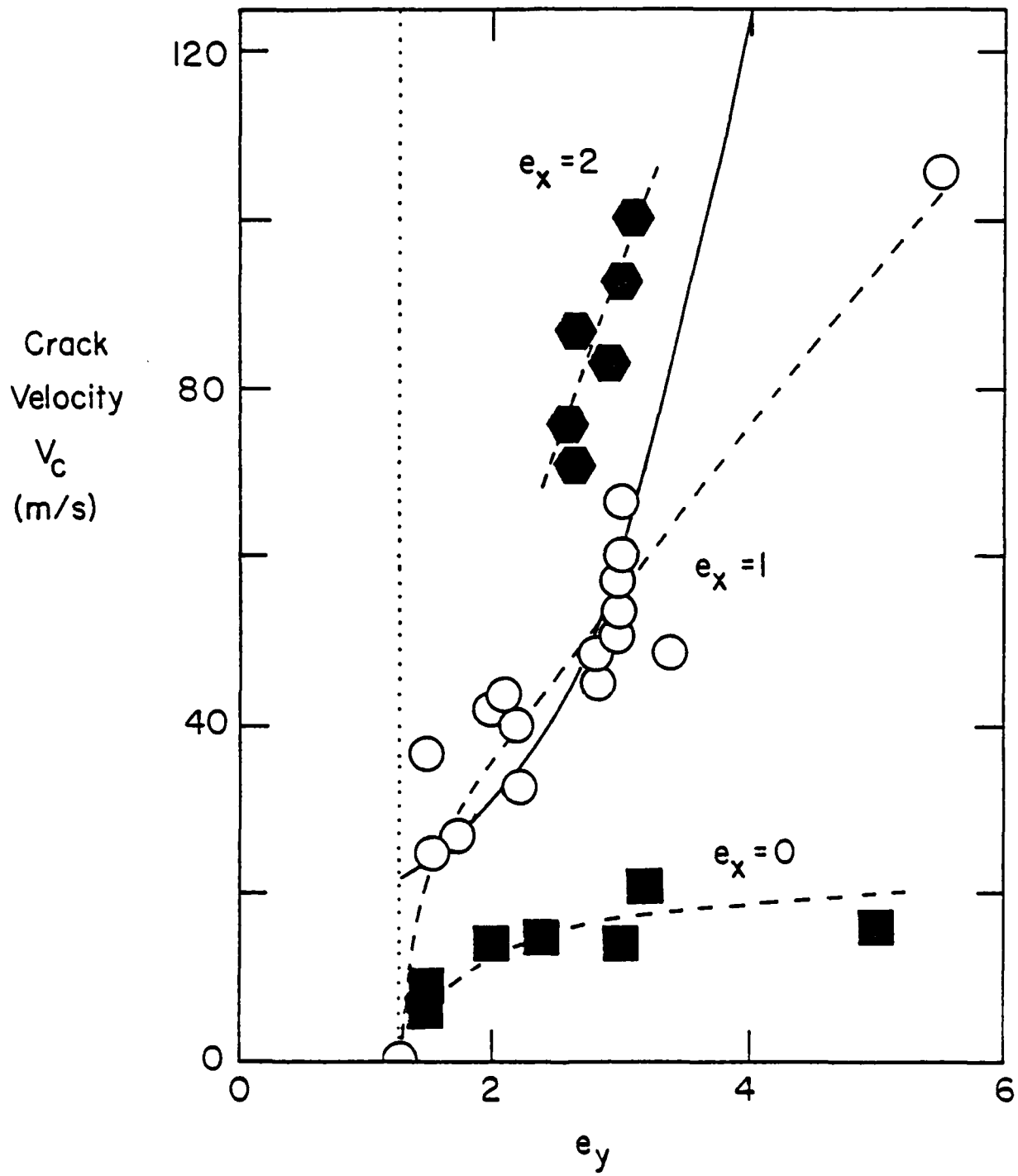
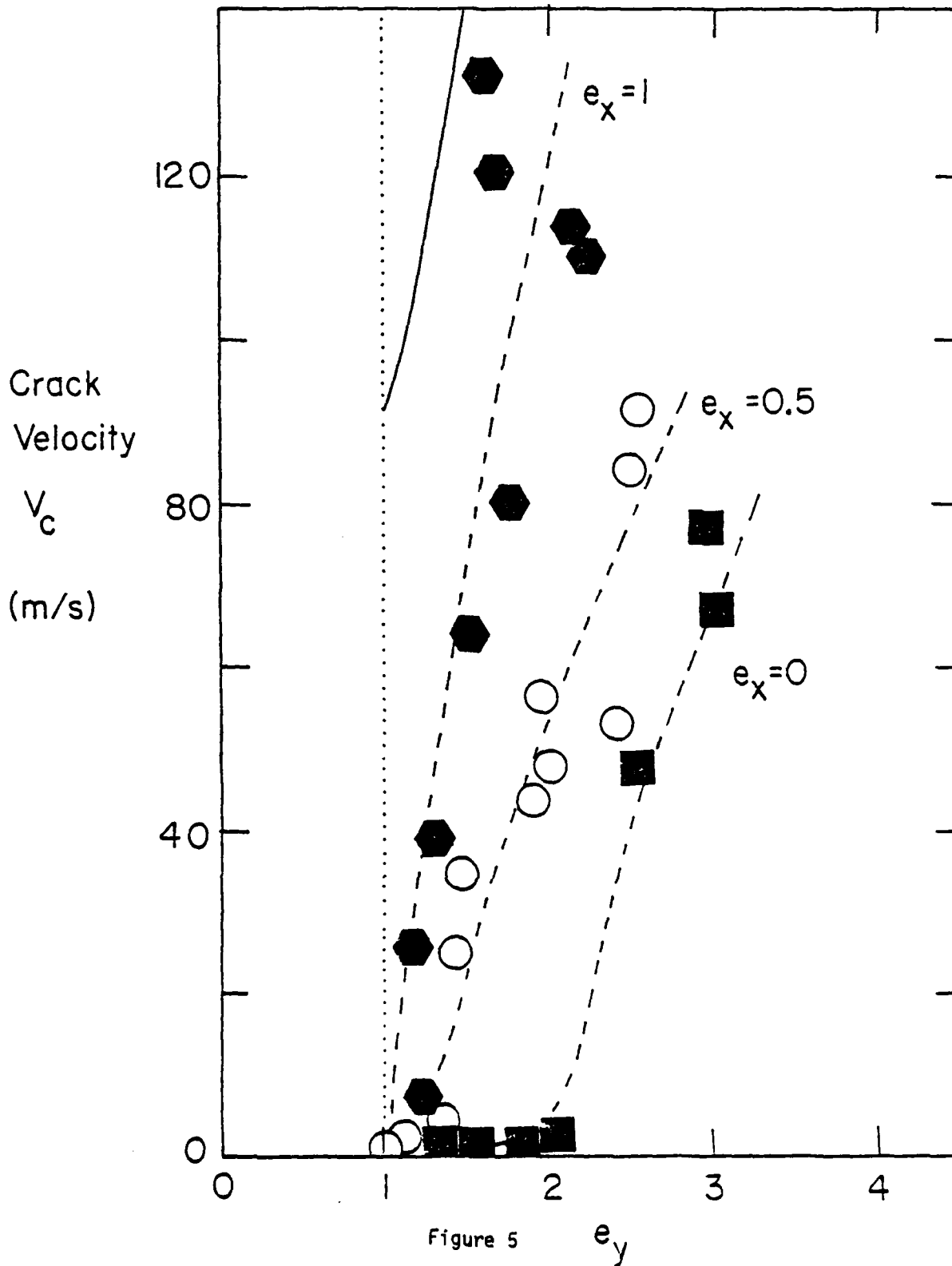


Figure 4





DYN

23  
6/61

DISTRIBUTION LIST

	<u>No. Copies</u>		<u>No. Copies</u>
Dr. L.V. Schmidt Assistant Secretary of the Navy (R,E, and S) Room 5E 731 Pentagon Washington, D.C. 20350	1	Dr. F. Roberto Code AFRPL MKPA Edwards AFB, CA 93523	1
Dr. A.L. Slafkosky Scientific Advisor Commandant of the Marine Corps Code RD-1 Washington, D.C. 20380	1	Dr. L.H. Caveny Air Force Office of Scientific Research Directorate of Aerospace Sciences Bolling Air Force Base Washington, D.C. 20332	1
Dr. Richard S. Miller Office of Naval Research Code 413 Arlington, VA 22217	10	Mr. Donald L. Ball Air Force Office of Scientific Research Directorate of Chemical Sciences Bolling Air Force Base Washington, D.C. 20332	1
Mr. David Siegel Office of Naval Research Code 250 Arlington, VA 22217	1	Dr. John S. Wilkes, Jr. FJSRL/NC USAF Academy, CO 80840	1
Dr. R.J. Marcus Office of Naval Research Western Office 1030 East Green Street Pasadena, CA 91106	1	Dr. R.L. Lou Aerojet Strategic Propulsion Co. P.O. Box 15699C Sacramento, CA 95813	1
Dr. Larry Peebles Office of Naval Research East Central Regional Office 666 Summer Street, Bldg. 114-D Boston, MA 02210	1	Dr. V.J. Keenan Anal-Syn Lab Inc. P.O. Box 547 Paoli, PA 19301	1
Dr. Phillip A. Miller Office of Naval Research San Francisco Area Office One Hallidie Plaza, Suite 601 San Francisco, CA 94102	1	Dr. Philip Howe Army Ballistic Research Labs ARRADCOM Code DRDAR-BLT Aberdeen Proving Ground, MD 21005	1
Mr. Otto K. Heiney AFATL - DLDL Elgin AFB, FL 32542	1	Mr. L.A. Watermeier Army Ballistic Research Labs ARRADCOM Code DRDAR-BLI Aberdeen Proving Ground, MD 21005	1
Mr. R. Geisler ATTN: MKP/MS24 AFRPL Edwards AFB, CA 93523	1	Dr. W.W. Wharton Attn: DRSMI-RKL Commander U.S. Army Missile Command Redstone Arsenal, AL 35898	1

## DISTRIBUTION LIST

	<u>No. Copies</u>		<u>No. Copies</u>
Dr. R.G. Rhoades Commander Army Missile Command DRSMI-R Redstone Arsenal, AL 35898	1	Dr. E.H. Debutts Hercules Inc. Baccus Works P.O. Box 98 Magna, UT 84044	1
Dr. W.D. Stephens Atlantic Research Corp. Pine Ridge Plant 7511 Wellington Rd. Gainesville, VA 22065	1	Dr. James H. Thacher Hercules Inc. Magna Baccus Works P.O. Box 98 Magna, UT 84044	1
Dr. A.W. Barrows Ballistic Research Laboratory USA ARRADCOM DRDAR-BLP Aberdeen Proving Ground, MD 21005	1	Mr. Theodore M. Gilliland Johns Hopkins University APL Chemical Propulsion Info. Agency Johns Hopkins Road Laurel, MD 20810	1
Dr. C.M. Frey Chemical Systems Division P.O. Box 358 Sunnyvale, CA 94086	1	Dr. R. McGuire Lawrence Livermore Laboratory University of California Code L-324 Livermore, CA 94550	1
Professor F. Rodriguez Cornell University School of Chemical Engineering Olin Hall, Ithaca, N.Y. 14853	1	Dr. Jack Linsk Lockheed Missiles & Space Co. P.O. Box 504 Code Org. 83-10, Bldg. 154 Sunnyvale, CA 94088	1
Defense Technical Information Center DTIC-DDA-2 Cameron Station Alexandria, VA 22314	12	Dr. B.G. Craig Los Alamos National Lab P.O. Box 1663 NSP/DOD, MS-245 Los Alamos, NM 87545	1
Dr. Rocco C. Musso Hercules Aerospace Division Hercules Incorporated Alleghany Ballistic Lab P.O. Box 210 Washington, D.C. 21502	1	Dr. R.L. Rabie WX-2, MS-952 Los Alamos National Lab. P.O. Box 1663 Los Alamos NM 37545	1
Dr. Ronald L. Simmons Hercules Inc. Eglin AFATL/DLCL Eglin AFB, FL 32542	1	Dr. B. Rogers Los Alamos Scientific Lab. P.O. Box 1663 Los Alamos, NM 87545	1

DYN

25  
6/81

## DISTRIBUTION LIST

	<u>No. Copies</u>		<u>No. Copies</u>
Mr. R. Brown Naval Air Systems Command Code 330 Washington, D.C. 20361	1	Dr. J. Schnur Naval Research Lab. Code 6510 Washington, D.C. 20375	1
Dr. H. Rosenwasser Naval Air Systems Command AIR-310C Washington, D.C. 20360	1	Mr. R. Beauregard Naval Sea Systems Command SEA 64E Washington, D.C. 20362	1
Mr. B. Sobers Naval Air Systems Command Code 03P25 Washington, D.C. 20360	1	Mr. G. Edwards Naval Sea Systems Command Code 62R3 Washington, D.C. 20362	1
Dr. L.R. Rothstein Assistant Director Naval Explosives Dev. Engineering Dept. Naval Weapons Station Yorktown, VA 23691	1	Mr. John Boyle Materials Branch Naval Ship Engineering Center Philadelphia, PA 19112	1
Dr. Lionel Dickinson Naval Explosive Ordnance Disposal Tech. Center Code D Indian Head, MD 20640	1	Dr. H.G. Adolph Naval Surface Weapons Center Code R11 White Oak Silver Spring, MD 20910	1
Mr. C.L. Adams Naval Ordnance Station Code PM4 Indian Head, MD 20640	1	Dr. T.D. Austin Naval Surface Weapons Center Code R16 Indian Head, MD 20640	1
Mr. S. Mitchell Naval Ordnance Station Code 5253 Indian Head, MD 20640	1	Dr. T. Hall Code R-11 Naval Surface Weapons Center White Oak Laboratory Silver Spring, MD 20910	1
Dr. William Tolles Dean of Research Naval Postgraduate School Monterey, CA 93940	1	Mr. G.L. Mackenzie Naval Surface Weapons Center Code R101 Indian Head, MD 20640	1
Naval Research Lab. Code 6100 Washington, D.C. 20375	1	Dr. K.F. Mueller Naval Surface Weapons Center Code R11 White Oak Silver Spring, MD 20910	1

## DISTRIBUTION LIST

	<u>No. Copies</u>		<u>No. Copies</u>
Mr. J. Murrin Naval Sea Systems Command Code 62R2 Washington, D.C. 20362	1	Dr. A. Nielsen Naval Weapons Center Code 385 China Lake, CA 93555	1
Dr. D.J. Pastine Naval Surface Weapons Center Code R04 White Oak Silver Spring, MD 20910	1	Dr. R. Reed, Jr. Naval Weapons Center Code 388 China Lake, CA 93555	1
Mr. L. Roslund Naval Surface Weapons Center Code R122 White Oak, Silver Spring MD 20910	1	Dr. L. Smith Naval Weapons Center Code 3205 China Lake, CA 93555	1
Mr. M. Stosz Naval Surface Weapons Center Code R121 White Oak Silver Spring, MD 20910	1	Dr. B. Douda Naval Weapons Support Center Code 5042 Crane, Indiana 47522	1
Dr. E. Zimmet Naval Surface Weapons Center Code R13 White Oak Silver Spring, MD 20910	1	Dr. A. Faulstich Chief of Naval Technology MAT Code 0716 Washington, D.C. 20360	1
Dr. D. R. Derr Naval Weapons Center Code 388 China Lake, CA 93555	1	LCDR J. Walker Chief of Naval Material Office of Naval Technology MAT, Code 0712 Washington, D.C. 20360	1
Mr. Lee N. Gilbert Naval Weapons Center Code 3205 China Lake, CA 93555	1	Mr. Joe McCartney Naval Ocean Systems Center San Diego, CA 92152	1
Dr. E. Martin Naval Weapons Center Code 3858 China Lake, CA 93555	1	Dr. S. Yamamoto Marine Sciences Division Naval Ocean Systems Center San Diego, CA 91232	1
Mr. R. McCarten Naval Weapons Center Code 3272 China Lake, CA 93555	1	Dr. G. Bosmajian Applied Chemistry Division Naval Ship Research & Development Center Annapolis, MD 21401	1
		Dr. H. Shuey Rohn and Haas Company Huntsville, Alabama 35801	1

## DISTRIBUTION LIST

<u>No. Copies</u>	<u>No. Copies</u>
Dr. J.F. Kincaid Strategic Systems Project Office Department of the Navy Room 901 Washington, D.C. 20376	1
Strategic Systems Project Office Propulsion Unit Code SP2701 Department of the Navy Washington, D.C. 20376	1
Mr. E.L. Throckmorton Strategic Systems Project Office Department of the Navy Room 1048 Washington, D.C. 20376	1
Dr. D.A. Flanigan Thickol Huntsville Division Huntsville, Alabama 35807	1
Mr. G.F. Mangum Thickol Corporation Huntsville Division Huntsville, Alabama 35807	1
Mr. E.S. Sutton Thickol Corporation Elkton Division P.O. Box 241 Elkton, MD 21921	1
Dr. G. Thompson Thickol Wasatch Division MS 240 P.O. Box 524 Brigham City, UT 84302	1
Dr. T.F. Davidson Technical Director Thickol Corporation Government Systems Group P.O. Box 9253 Ogden, Utah 84409	1
Dr. C.W. Vriesen Thickol Elkton Division P.O. Box 241 Elkton, MD 21921	1
Dr. J.C. Hinshaw Thickol Wasatch Division P.O. Box 524 Brigham City, Utah 84302	1
U.S. Army Research Office Chemical & Biological Sciences Division P.O. Box 12211 Research Triangle Park NC 27709	1
Dr. R.F. Walker USA ARRADCOM DRDAR-LCE Dover, NJ 07801	1
Dr. T. Sinden Munitions Directorate Propellants and Explosives Defence Equipment Staff British Embassy 3100 Massachusetts Ave. Washington, D.C. 20008	1
LTC B. Loving AFROL/LK Edwards AFB, CA 93523	1
Professor Alan N. Gent Institute of Polymer Science University of Akron Akron, OH 44325	1
Mr. J. M. Frankle Army Ballistic Research Labs ARRADCOM Code DRDAR-BLI Aberdeen Proving Ground, MD 21005	1

DYN

28

6/81

## DISTRIBUTION LIST

<u>No. Copies</u>		<u>No. Copies</u>	
1	Dr. Ingo W. May Army Ballistic Research Labs ARRADCOM Code DRDAR-BLI Aberdeen Proving Ground, MD 21005	1	Dr. J. P. Marshall Dept. 52-35, Bldg. 204/2 Lockheed Missile & Space Co. 3251 Hanover Street Palo Alto, CA 94304
1	Professor N.W. Tschoegl California Institute of Tech Dept. of Chemical Engineering Pasadena, CA 91125	1	Ms. Joan L. Janney Los Alamos National Lab Mail Stop 920 Los Alamos, NM 87545
1	Professor M.D. Nicol University of California Dept. of Chemistry 405 Hilgard Avenue Los Angeles, CA 90024	1	Dr. J. M. Walsh Los Alamos Scientific Lab Los Alamos, NM 87545
1	Professor A. G. Evans University of California Berkeley, CA 94720	1	Professor R. W. Armstrong Univ. of Maryland Department of Mechanical Eng. College Park, MD 20742
1	Professor T. Litovitz Catholic Univ. of America Physics Department 520 Michigan Ave., N.E. Washington, D.C. 20017	1	Prof. Richard A. Reinhardt Naval Postgraduate School Physics & Chemistry Dept. Monterey, CA 93940
1	Professor W. G. Knauss Graduate Aeronautical Lab California Institute of Tech. Pasadena, CA 91125	1	Dr. R. Barnecker Naval Surface Weapons Center Code R13 White Oak, Silver Spring, MD 20910
1	Professor Edward Price Georgia Institute of Tech. School of Aerospace Engin. Atlanta, Georgia 30332	1	Dr. M. J. Kamlet Naval Surface Weapons Center Code R11 White Oak, Silver Spring, MD 20910
1	Dr. Kenneth O. Hartman Hercules Aerospace Division Hercules Incorporated P.O. Box 210 Cumberland, MD 21502	1	Professor J. D. Achenbach Northwestern University Dept. of Civil Engineering Evanston, IL 60201
1	Dr. Thor L. Smith IBM Research Lab 042.282 San Jose, CA 95193	1	Dr. N. L. Basdekas Office of Naval Research Mechanics Program, Code 432 Arlington, VA 22217
		1	Professor Kenneth Kuo Pennsylvania State Univ. Dept. of Mechanical Engineering University Park, PA 16802

DYN

20  
6/81

DISTRIBUTION LIST

	<u>No. Copies</u>	<u>No. Copies</u>
Dr. S. Sheffield Sandia Laboratories Division 2513 P.O. Box 5800 Albuquerque, NM 87185	1	
Dr. M. Farber Space Sciences, Inc. 135 Maple Avenue Monrovia, CA 91016	1	
Dr. Y. M. Gupta SRI International 333 Ravenswood Avenue Menlo Park, CA 94025	1	
Mr. M. Hill SRI International 333 Ravenswood Avenue Menlo Park, CA 94025	1	
Professor Richard A. Schapery Texas A&M Univ. Dept of Civil Engineering College Station, TX 77843	1	
Dr. Stephen Swanson Univ. of Utah Dept. of Mech. & Industrial Engineering MEB 3008 Salt Lake City, UT 84112	1	
Mr. J. D. Byrd Thiokol Corp. Huntsville Huntsville Div. Huntsville, AL 35807	1	
Professor G. D. Duvall Washington State University Dept. of Physics Pullman, WA 99163	1	
Prof. T. Dickinson Washington State University Dept. of Physics Pullman, WA 99163	1	



DATE  
ILME  
—8